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**FM Elsewhere:**  
**Analyzing Cockpit Interfaces**  
**Using Formal Methods**

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## Overview

- An “elsewhere” example:  
**Automation surprises in cockpit interfaces**
- Issues in working “elsewhere”
- What makes formal methods effective “elsewhere”?

## Aviation Background

- Modern passenger aircraft are very reliable
- The dominant cause of incidents and accidents is human error (70% of accidents)
- Modern cockpits are highly automated
  - And highly complicated
  - Can sometimes override the pilot
- Pilots can be surprised by the behavior of the automation
  - Or confused about what “mode” it is in
  - “Why did it do that?”
  - “What is it doing now?”
  - “What will it do next?”
- Can formal methods help?

## Postulates (from Human Factors)

- Operators use “mental models” to guide their interaction with automated systems
- Automation surprises arise when the operator’s mental model does not accurately reflect the behavior of the actual system
- Mode confusion is a just a special case: the mental model is not an accurate reflection of the actual mode structure
  - Or loses sync with it
- Mental models can be explicitly formulated as state machines
  - And we can “capture” them through observation, interviews, and introspection
  - Or by studying training manuals  
(which are intended to induce specific models)

## Facts (from Computer Science)

- The behavior of automated systems can be formulated in terms of (interacting) state machines
- These state machine descriptions are increasingly being used to document requirements and designs (cf. Statemate, UML)
- A technology called “model checking” can be used to examine the complete behavior of very large state machines
  - Can examine many millions of states
  - Used routinely in h/w design, s/w requirements analysis
  - It is largely automatic
- Can check whether certain properties are always true (e.g., every operator input is eventually acknowledged)
- Or can compare whether two state machines are “consistent”
- Produces counterexample when divergence found

## Putting These Together

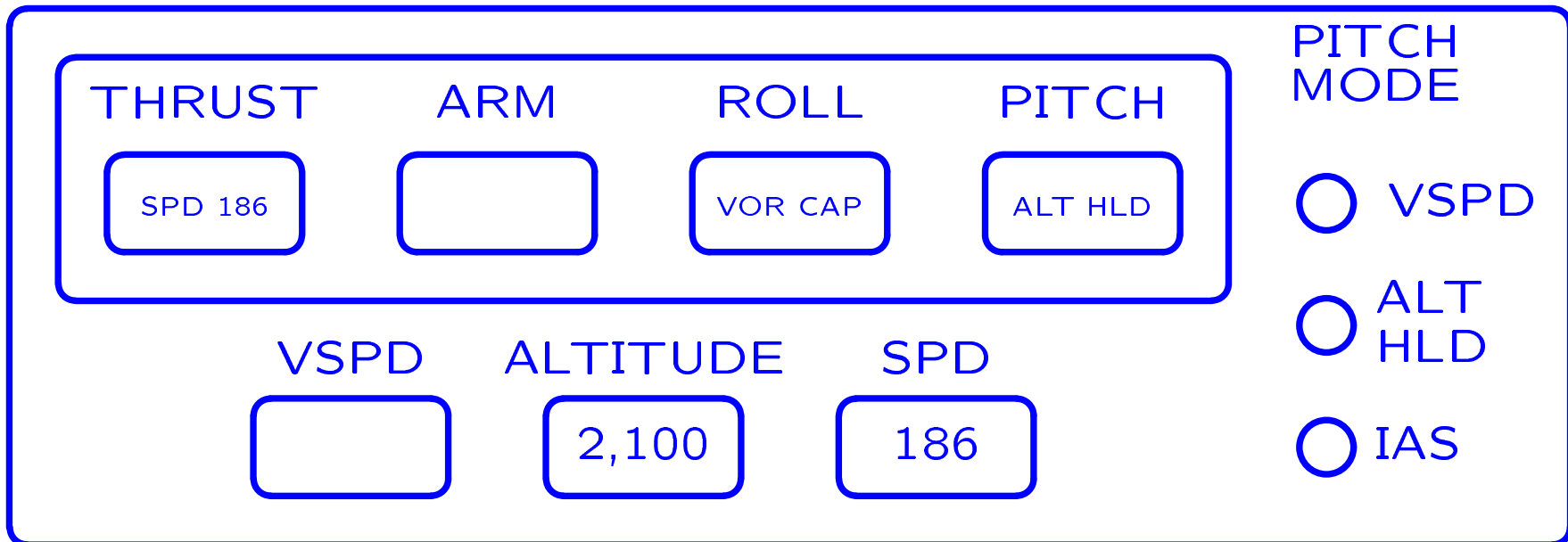
- Take the **design** of an automated system
  - Represented as a state machine
- And that of a (plausible or actual) **mental model**
  - Also represented as a state machine

And check them for consistency

- **Any counterexamples will be potential automation surprises**

## Example: Altitude Bust Scenario

- Scenario describes an automation surprise in the MD-88 autopilot (from Ev Palmer)
- Crew had just made a missed approach
- Climbed and leveled at 2,100 feet

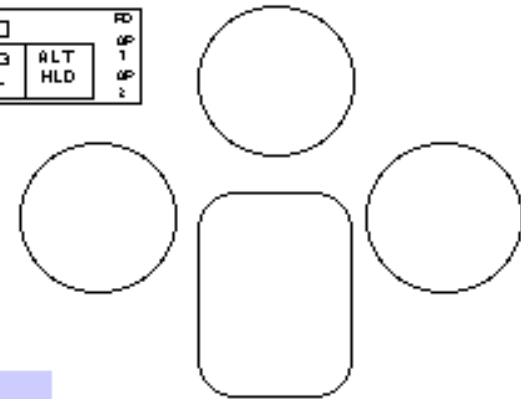
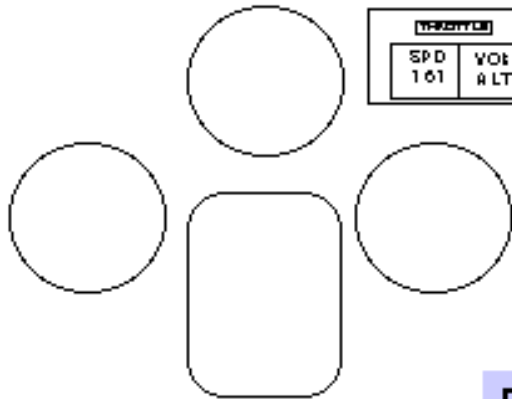
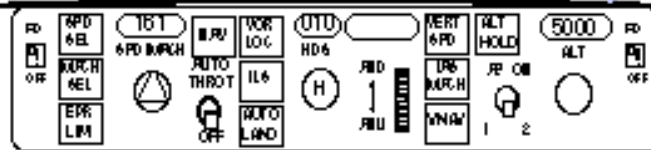


Color code: **done by pilot**, **done by others or by automation**



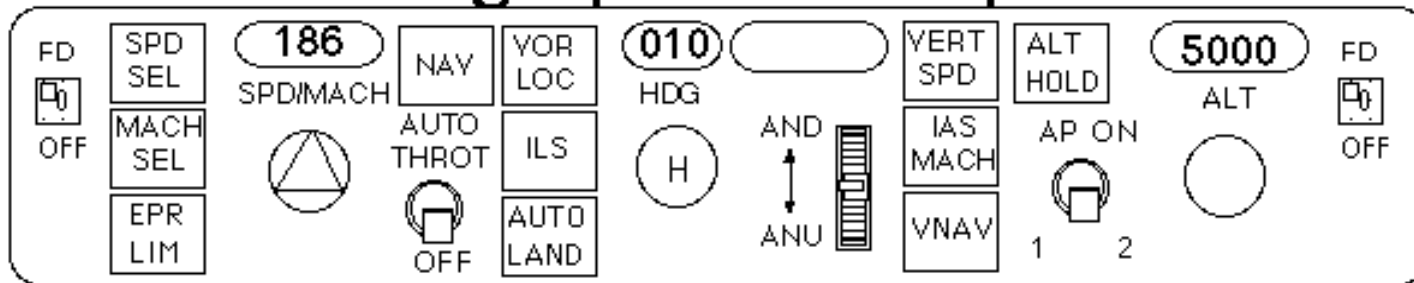
## A More Realistic Picture

Mode Control Panel (MCP)



Flight Mode Annunciator (FMA)

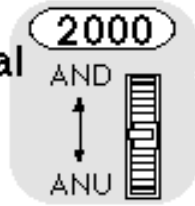
# Setting up the Autopilot



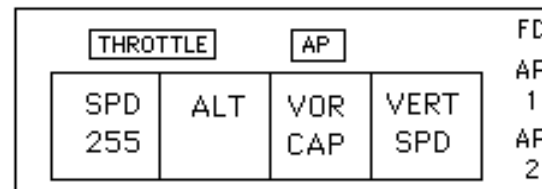
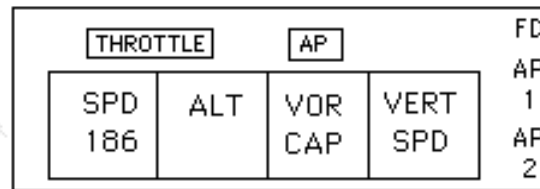
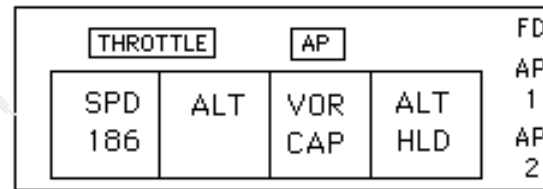
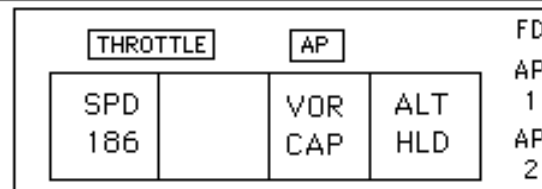
1. Set altitude to 5000



2. Set vertical speed to 2000



3. Set thrust to maintain a speed of 255.

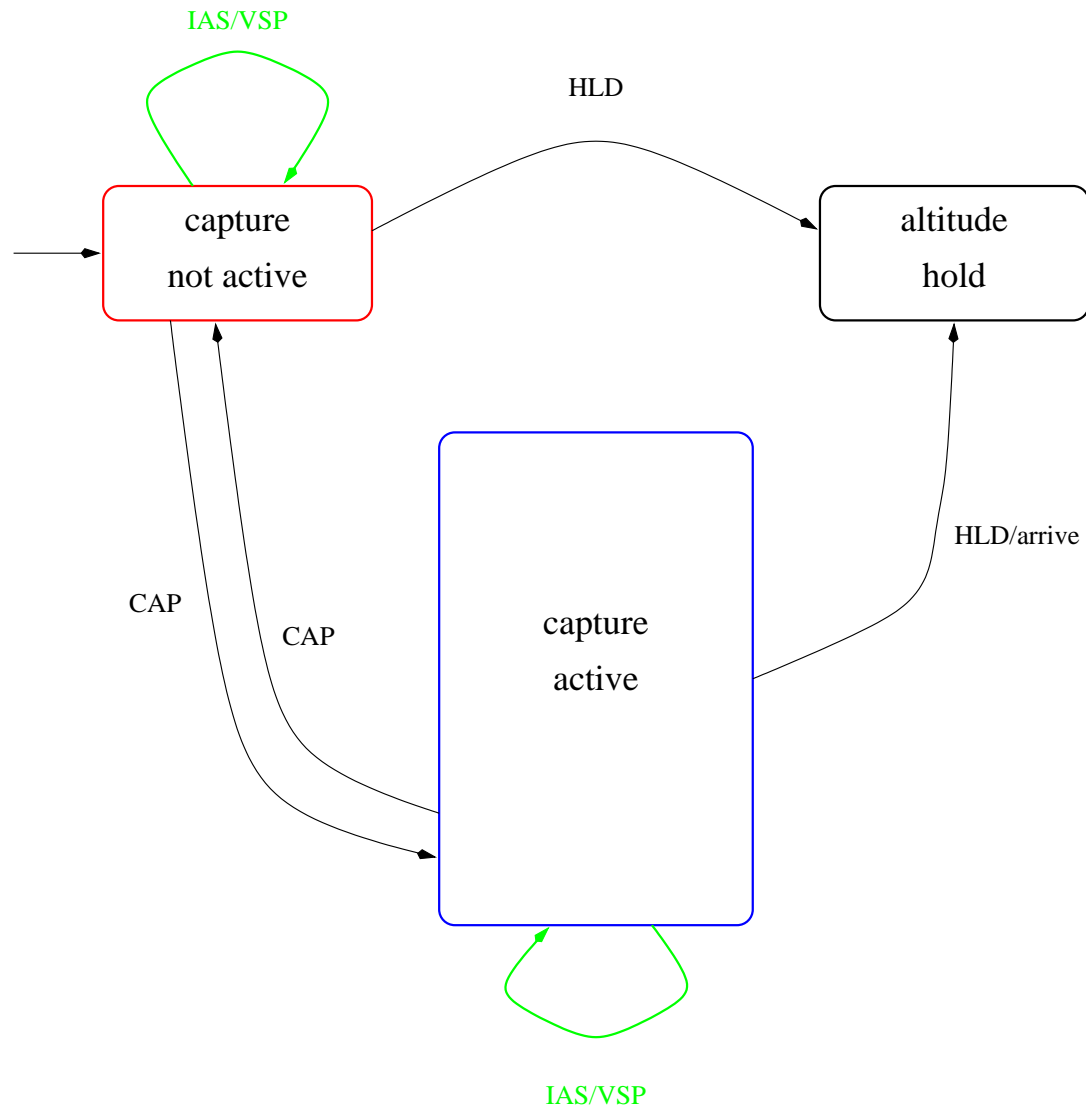


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## Altitude Bust Scenario: Mental Model

- The **pitch modes** determine **how** the plane climbs
  - **VSPD**: climb at so many feet per minute
  - **IAS**: climb while maintaining set airspeed
  - **ALT HLD**: hold current altitude
- The **altitude capture** mode determines whether there is a **limit** to the climb
  - If altitude capture is **armed**
    - ★ Plane will climb to set altitude and hold it
    - ★ There is also an **ALT CAP** pitch mode that is used to end the climb smoothly
  - Otherwise
    - ★ Plane will keep climbing until pilot stops it

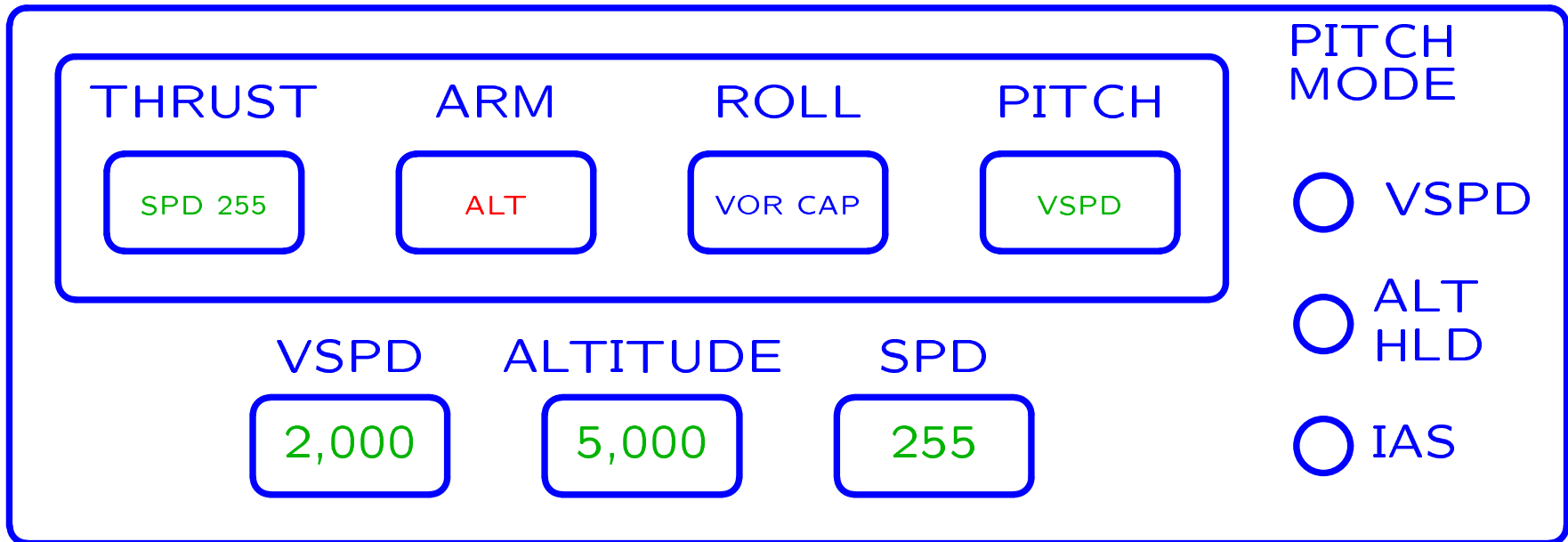
# Mental Model



Whether capture is active is independent of the pitch mode

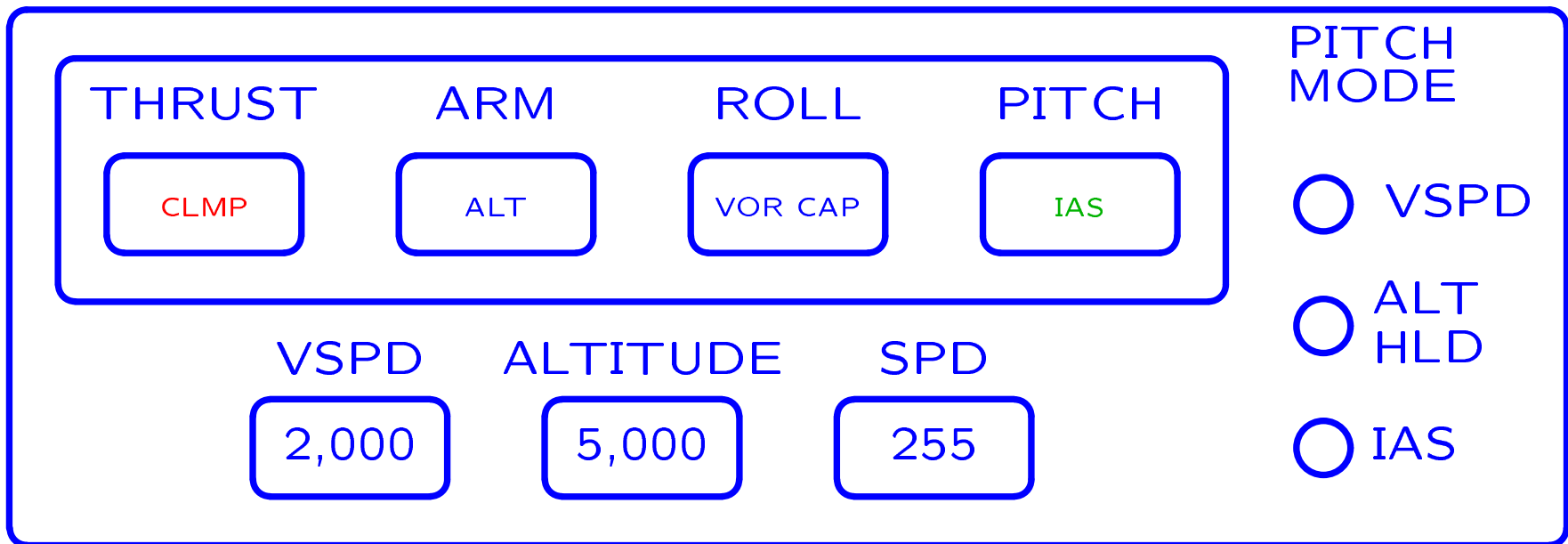
## Altitude Bust Scenario—II

- Air traffic Control: “Climb and maintain 5,000 feet”
- Captain set MCP altitude window to 5,000 feet
  - Causes ALT capture to arm
- Also set pitch mode to VSPD with a value of 2,000 fpm
- And autothrottle (thrust) to SPD mode at 255 knots



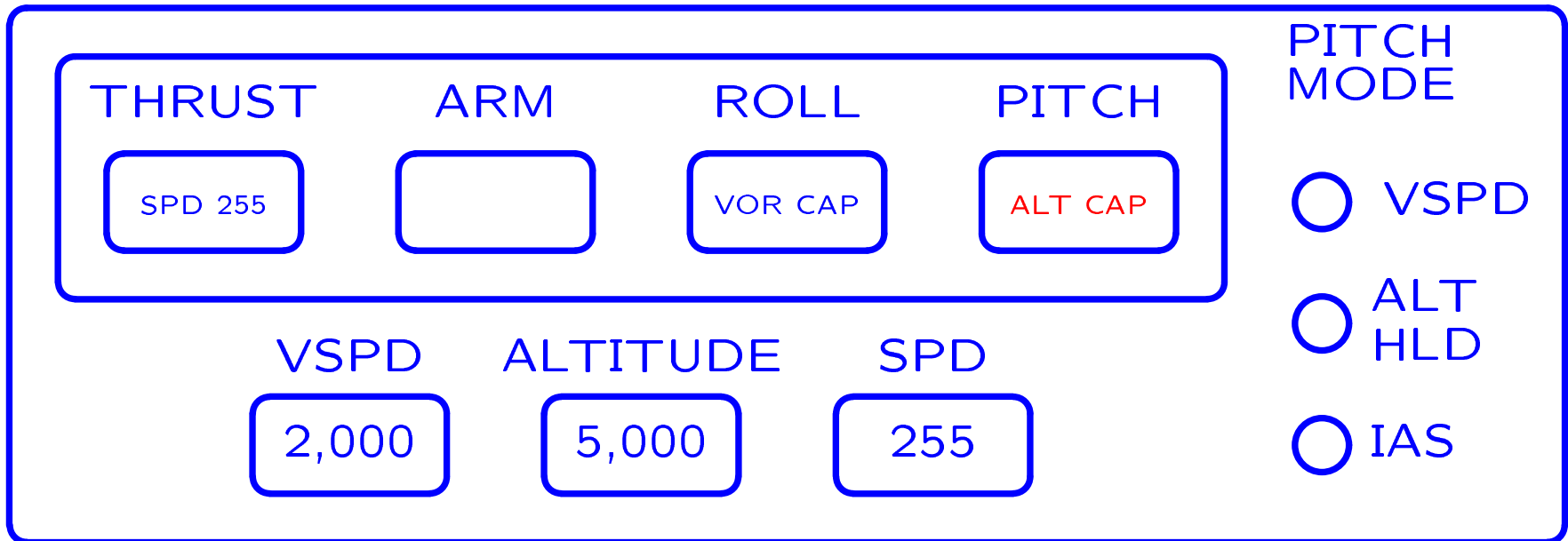
## Altitude Bust Scenario—III

- Climbing through 3,500 feet, flaps up, slats retract
- Captain changed pitch mode to IAS
  - Causes autothrottle (thrust) to go to CLMP



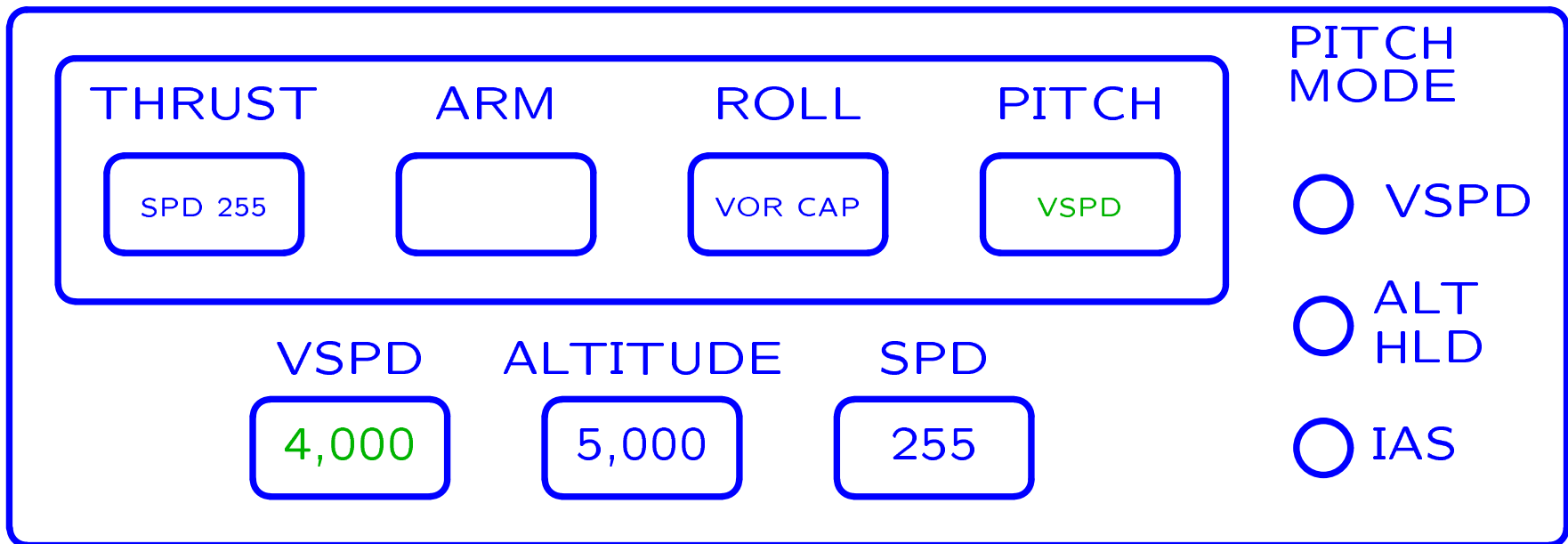
## Altitude Bust Scenario—IV

- Three seconds later, nearing 5,000 feet, autopilot automatically changed **pitch mode** to **ALT CAP**
  - And **disarmed ALT** capture



## Altitude Bust Scenario—V

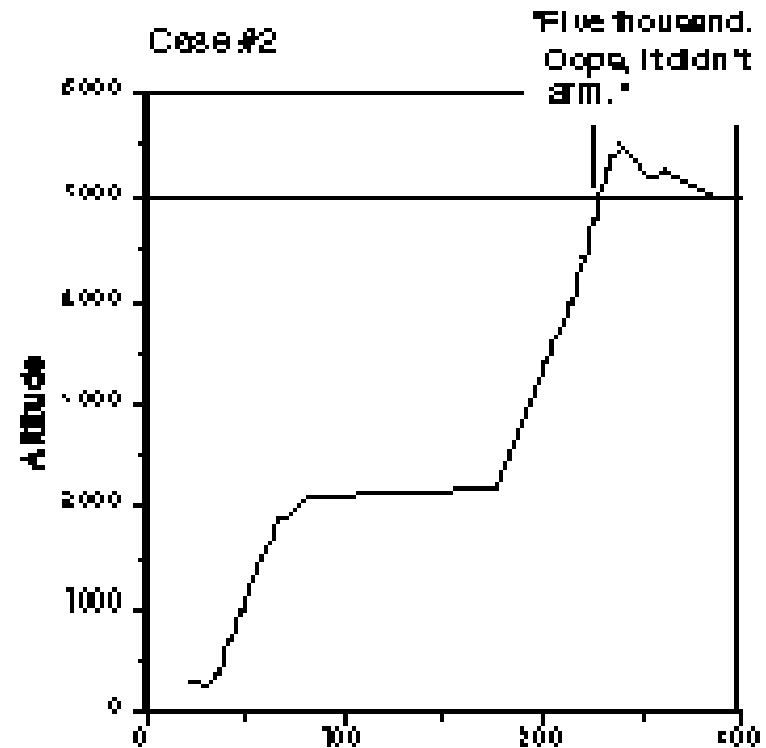
- 1/10 second later, Captain changed **VSPD** dial to **4,000** fpm





## Altitude Bust: Outcome

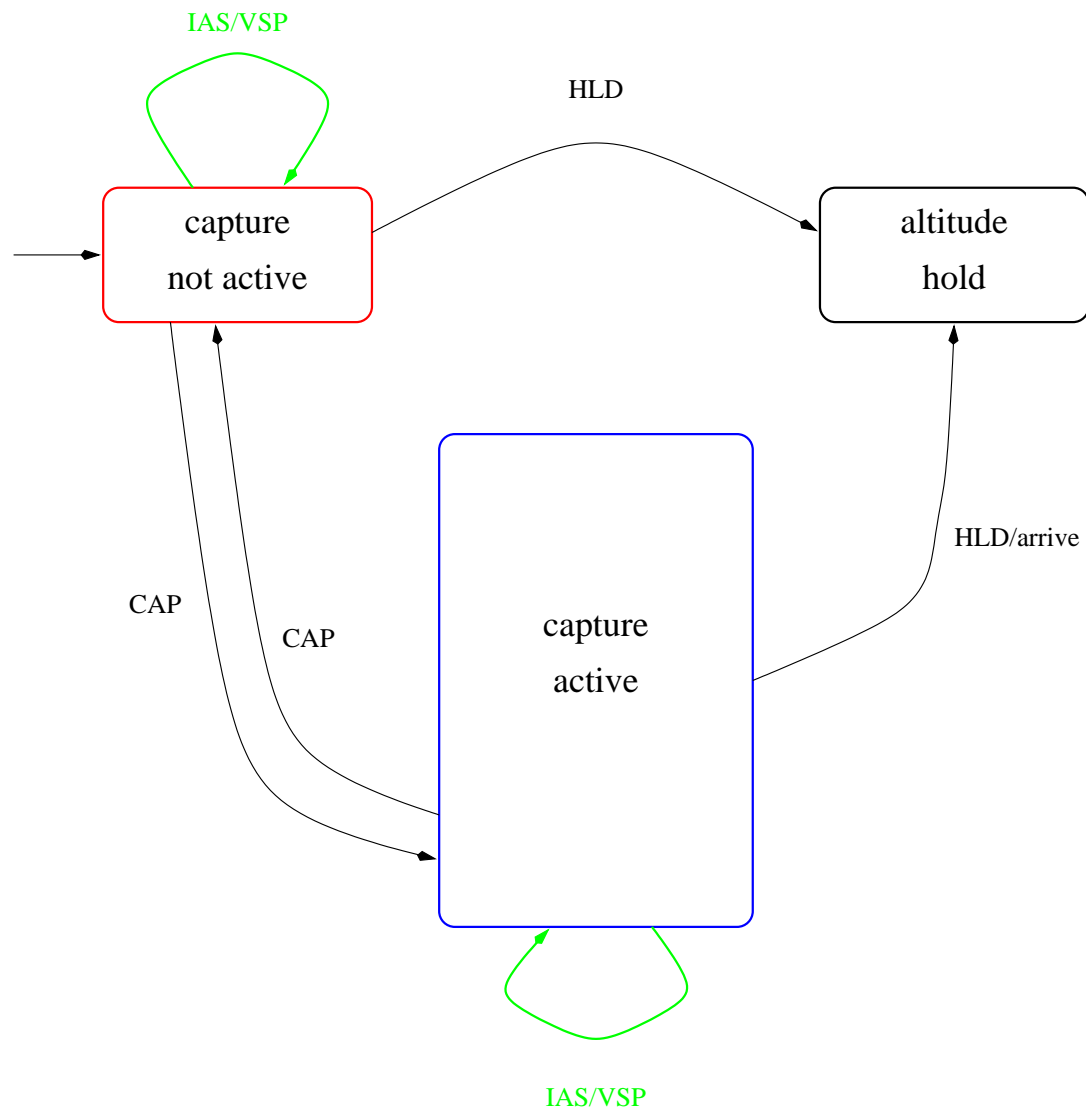
- Plane passed through 5,000 feet at vertical velocity of 4,000 fpm
- “Oops: It didn’t arm”
- Captain took manual control, halted climb at 5,500 with the “*altitude—altitude*” voice warning sounding repeatedly



## Automated Discovery of the Altitude Bust Scenario

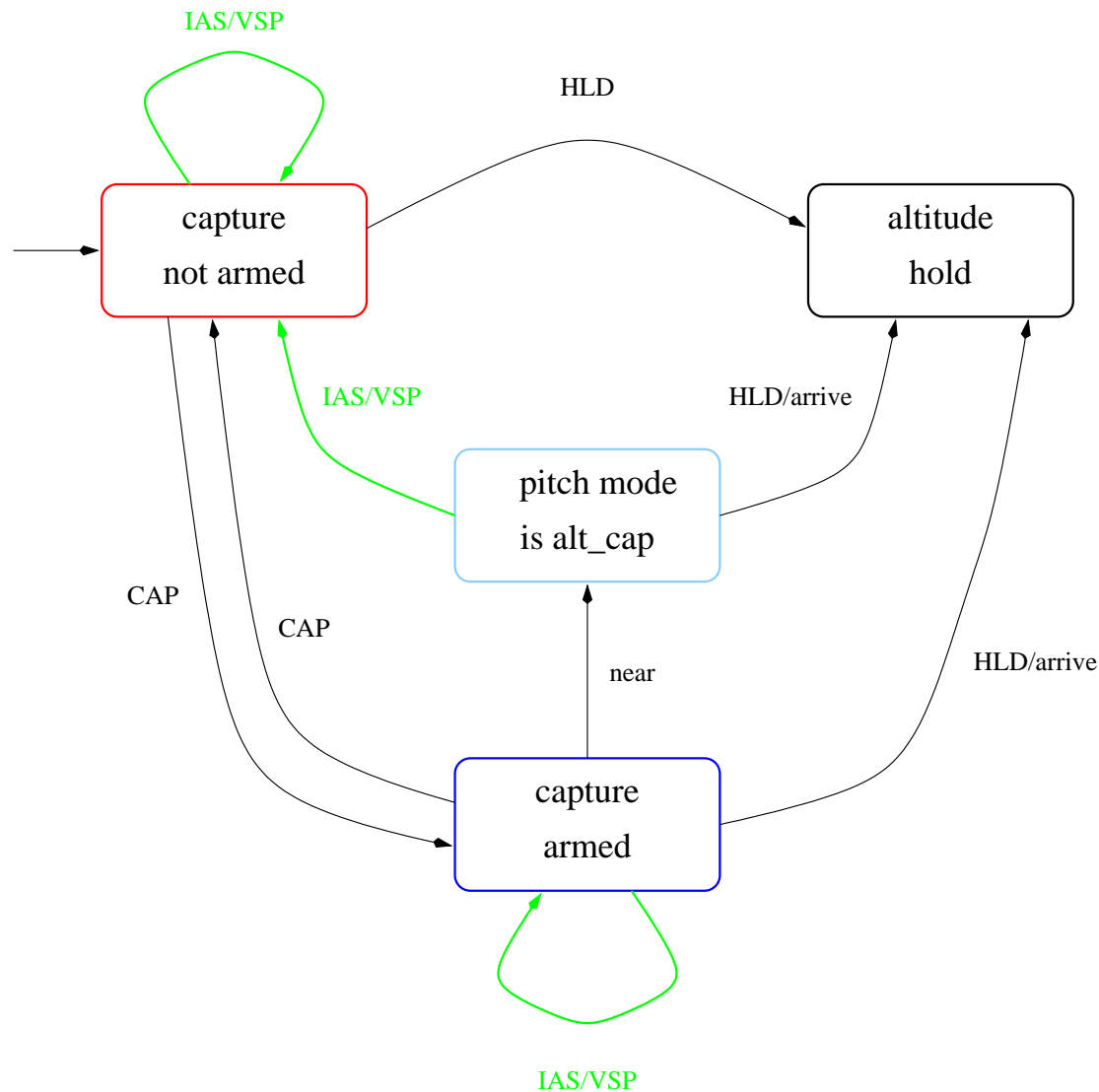
- I did it using a model checker called Mur $\phi$ 
  - Comes from David Dill's group at Stanford
- But first I'll explain it using diagrams

## Mental Model (again)



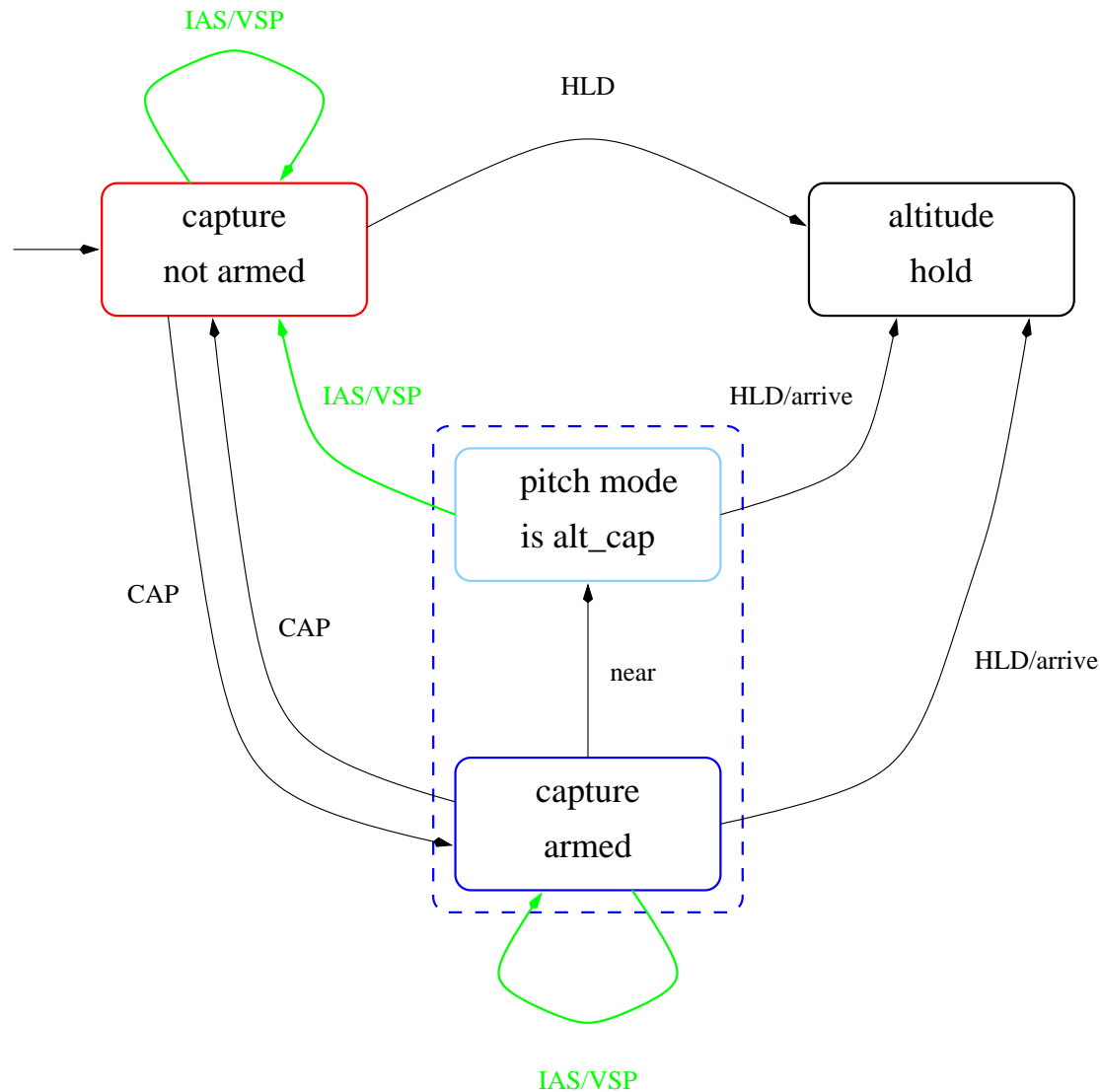
Whether capture is active is independent of the pitch mode

# Actual System



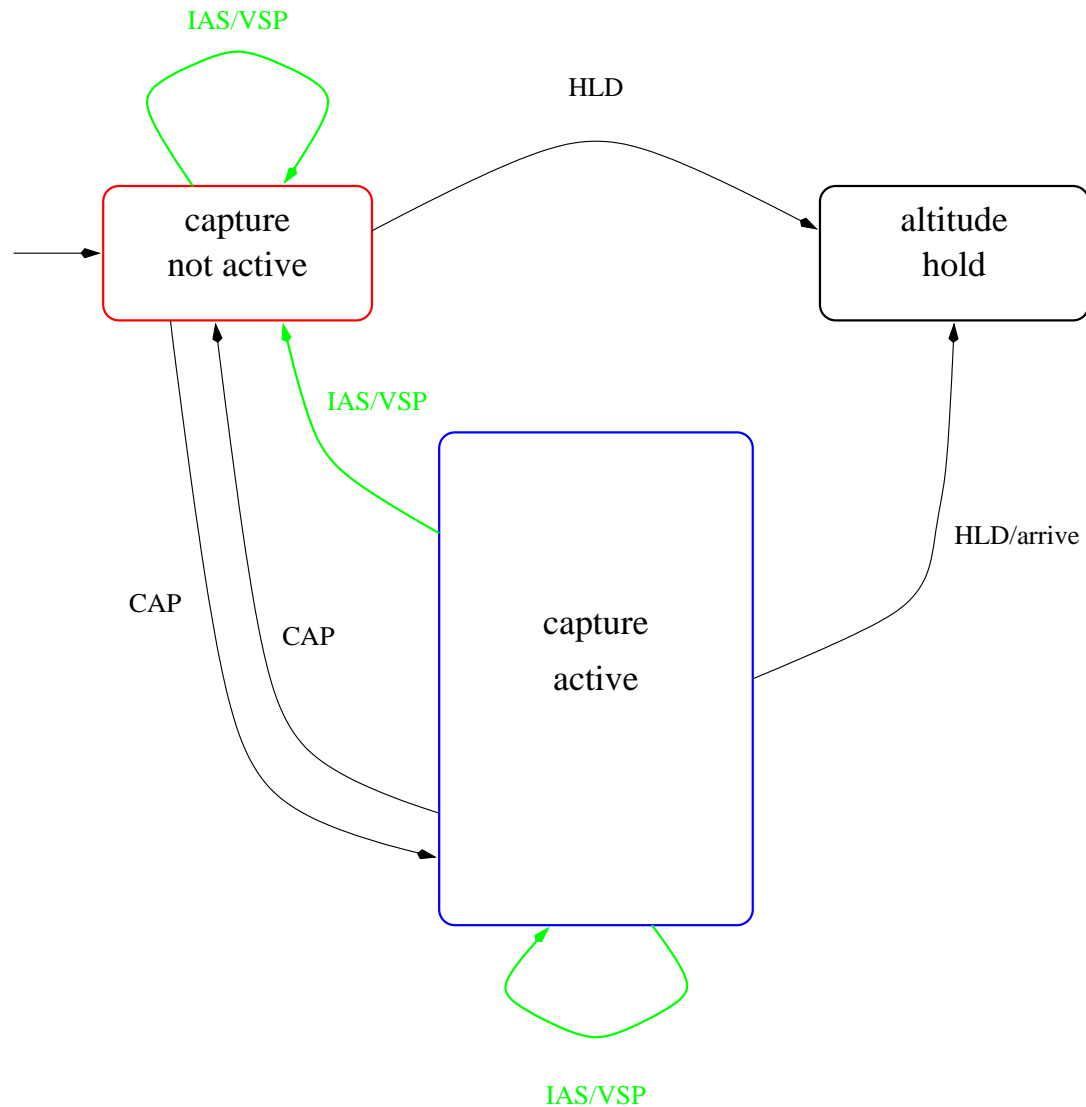
There is an alt\_cap pitch mode that flies the final capture

## Focus (Abstract) on Whether Capture Is Active



Capture is active if it is armed or if pitch mode is alt\_cap

# Abstracted System



Can compare this description directly with the mental model

# Altitude Bust: Mur $\phi$ Specification

```
type
  pitch_modes: enum{vert_speed, ias, alt_cap, alt_hold};
var
  pitch_mode: pitch_modes;
  capture_armed: boolean;
  ideal_capture: boolean;

startstate
begin
  clear pitch_mode;
  capture_armed := false;
  ideal_capture := false;
end;

rule "ALT CAPTURE"
begin
  capture_armed := !capture_armed;
  ideal_capture := !ideal_capture;
end;

rule "HLD"
begin
  pitch_mode := alt_hold;
end;

rule "IAS"
begin
  pitch_mode := ias;
end;

rule "VSPD"
begin
  pitch_mode := vert_speed;
end;

rule "near"
begin
  if capture_armed then
    capture_armed := false;
    pitch_mode := alt_cap;
  endif;
end;

rule "arrived"
begin
  if pitch_mode = alt_cap then
    pitch_mode := alt_hold;
  endif;
  if capture_armed then
    pitch_mode := alt_hold;
    capture_armed := false;
  endif;
  if ideal_capture then
    ideal_capture := false;
  endif;
end;

invariant ideal_capture = (capture_armed | pitch_mode = alt_cap)
```

## Altitude Bust: Mur $\phi$ Analysis

Invariant "Invariant 0" failed.

Startstate Startstate 0 fired.

pitch\_mode:vert\_speed

capture\_armed:false

ideal\_capture:false

-----

Rule ALT CAPTURE fired.

capture\_armed:true

ideal\_capture:true

-----

Rule near fired.

pitch\_mode:alt\_cap

capture\_armed:false

-----

Rule VSPD fired.

The last state of the trace (in full) is:

pitch\_mode:vert\_speed

capture\_armed:**false**

ideal\_capture:**true**



## Altitude Bust: Results

- Found the “surprise” scenario (in 0.24 seconds)
- So did Leveson and Palmer
  - By looking for “indirect mode changes”
- They suggested a fix (see HESSD paper)
- I incorporated it in my model
- And found that it caused another surprise
- I fixed that
- And found yet another surprise
  - (also present, in a different form, in original specification)
- I fixed that, and the system and the mental model now align

## Altitude Bust: Additional Experiment

- Mode confusions can arise even with consistent models if operator loses sync
- I introduced a rule to model a forgetful operator (nondeterministically flips the mental state)
- Obviously this introduces mode confusions
- I then modified the mental model to “reload” its state from a display that indicates whether altitude capture is armed
- This works (no surprises), even with a forgetful operator
- **Can be used to validate cues provided by displays**
  - Cf. Surprise in 737 autopilot

## Observations

- Once the initial model was constructed, these experiments required negligible effort (and only seconds of machine time)
- Provides complete demonstration of consistent behavior
  - Relative to the models used
  - General experience with model checking is that you learn more by examining all possibilities of a simplified model than by probing some of the possibilities of the full thing (cf. simulation or testing)
- Approach does not supplant the contributions of those working in human factors and aviation psychology
  - Provides a tool to examine properties of their models using automated calculation

## Comparisons

- **Leveson** enumerates error-prone design elements (e.g. indirect mode transitions)
  - And examines system design to locate them
    - ★ Must then determine whether those found are real problems in their specific context
  - Examination is not automated
  - Tension between examining too much and too little
- **Butler** (NASA Langley), **Miller** (Collins) and colleagues use mechanized formal methods (theorem proving and model checking) to examine specification of autopilot for safety invariants (e.g., no mode change without pilot input)
  - Similar to my approach
  - But mental model is richer specification than an invariant

## Further Work (TBD)

- Have also used this approach to examine a surprise related to speed protection in A320
  - Will also try it on a known surprise in 737 autopilot
- Need to try it out on large, realistic examples
- Denis Javaux (psychologist from University of Liège in Belgium) has proposed two processes that give mental models their “shape”
  - Could take the model implied by training manual, then apply these two simplification processes, to generate plausible mental models “automatically”
- Could also take mental model from one airplane and compare it to the automation from another as a way of predicting training difficulties

## Speculation

- Can also do **design exploration** on effects of
    - Simpler design
    - New operating instructions
    - Improved displays
    - Faulty operator
  - The mental model could also be interpreted as a requirements specification
    - Describes desired rather than observed operator interface
  - Lack of an **accurate** and **simple** mental model then suggests overly-complex design
    - How many states are needed?
    - Any complex data structures (e.g., a stack)?
- Minimal safe model assesses cognitive load**

## Technical Challenges: Methodological

Can only go so far modeling just the mode behavior

And abstracting everything else away

- Need to investigate incorporating limited models of the environment and of the control behavior
  - E.g., to distinguish climbing from descending, up from down
  - Qualitative physics may prove adequate
    - ★ Reasons about signs of quantities and rates of change
    - ★ E.g., climb means height increases (derivative is +)
  - May need hybrid automata (and model checkers for these)
- Also need to look at real time issues (e.g., delay between reading display and taking action)

## Issues In Working “Elsewhere”

- Obviously, need to learn something about another field
- Also need to learn how to talk to practitioners in another field
  - Pilots do not like being described as finite automata
  - Many psychologists don't have computational intuition
- In both cases, they see a “human factors” issue where we see a design problem



## When Are Formal Methods Effective “Elsewhere” ?

- Formal methods is simply mathematical modeling and analysis applied to logical systems
- Just like the use of mathematical modeling in other engineering disciplines (e.g. finite-elements analysis)
  - Only useful when mechanized
- Need fairly generic languages and tools
  - Those with commitment to a particular computational model, methodology, or other dogma may prove ineffective away from their home turf
- Can look forward to ubiquitous formal methods

## To Learn More (About Cockpit Automation)

- Our papers and technical reports are at <http://www.csl.sri.com/fm.html>
  - <http://www.csl.sri.com/~rushby/hessd99.html> describes this work and provides the Mur $\phi$  code
    - ★ Links to Mur $\phi$  there also
  - <http://www.csl.sri.com/~rushby/dasc99.html> and
  - <http://www.csl.sri.com/~rushby/hci-aero00.html> are other papers on this topic
- Information about our verification system, PVS, and the system itself are available from <http://www.csl.sri.com/pvs.html>
  - Runs under SunOS, Solaris, or RH (X86) Linux
  - Freely available under license to SRI